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ON THE ORIGINAL ORBIT OF
COMET 1899 I

BY

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Comet 1899 I belongs to the group of comets for which a definitive calculation gives a hyperbolic orbit and therefore is fit to be used as basis of an investigation of the original orbit before the comet entered the region of the Sun and the major planets. The following elements have been computed by Merfield from 580 observations from March 4 to August 10, 1899 (159 days), all perturbations being taken into account (*Astronomische Nachrichten* 3748).

Osculation 1899, March 12

$$\begin{array}{r}
 T = 1899, \text{ April } 12.978010 \text{ G.M.T.} \\
 \omega = 8^{\circ}41'46''.48 \\
 \Omega = 24 \ 59 \ 59.93 \\
 i = 146 \ 15 \ 30.29 \\
 q = 0.32657237 \pm 0.00000162 \\
 e = 1.00035029 \pm 0.00000404
 \end{array}
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 \end{array}
 \right\}
 \begin{array}{l}
 1900.0 \\
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 \\
 \text{Probable errors.}
 \end{array}$$

From the values of q and e we compute the reciprocal value of the semi-major axis and its mean error:—

$$\frac{1}{a} = -0.0010726 \pm 0.0000184.$$

The orbit does not quite fulfil Elis Strömgren's requirement, not having a period of observations of at least 6 months. As the number of observations, however, is large and as moreover another computation by Wedemeyer has given practically the same elements, I have all the same carried out a computation of the perturbed orbit in the years before the time of perihelion.

As the distance of perihelion is small, we have to start the computation by Encke's method. If we take all perturbations into account we get the following rectangular, ecliptical perturbations in units of the 8th decimal referred to the equinox of 1900.0:—

G. M. T.		ξ	η	ζ
1899 Mar.	30....	+ 32	0	+ 18
	20....	7	0	4
	10....	1	0	1
Feb.	28....	15	— 5	15
	18....	54	31	58
	8....	123	87	135
Jan.	29....	228	182	246
	19....	367	322	389
	9....	537	518	563
1898 Dec.	30....	730	778	769
	20....	941	1111	1012
Nov.	30....	1377	2022	1613
	10....	1848	3299	2400
Oct.	21....	2413	4971	3414
	1....	3112	7061	4707
Sep.	11....	3968	9596	6340
Aug.	22....	4983	12616	8388
	2....	6142	16181	10940
July	13....	7431	20374	14106
June	23....	+ 8832	— 25296	+ 18018

On July 13, 1898, the comet had a distance from the Sun sufficient for the direct integration of the co-ordinates. From the elements we compute the unperturbed co-ordinates and velocities:—

$$x_0 = - 3.9804483 \quad 20 \frac{dx_0}{dt} = + 0.22685216$$

$$y_0 = + 0.7341700 \quad 20 \frac{dy_0}{dt} = + 0.01624977$$

$$z_0 = - 1.5681124 \quad 20 \frac{dz_0}{dt} = + 0.05420179$$

From the scheme of perturbations we get:—

$$\xi = + 0.0000743 \quad 20 \frac{d\xi}{dt} = - 0.00001342$$

$$\eta = -0.0002037 \quad 20 \frac{d\eta}{dt} = +0.00004540$$

$$\xi = +0.0001411 \quad 20 \frac{d\xi}{dt} = -0.00003515$$

The additions give the perturbed co-ordinates and velocities:—

$$x = -3.9803740 \quad 20 \frac{dx}{dt} = +0.22683874$$

$$y = +0.7339663 \quad 20 \frac{dy}{dt} = +0.01629517$$

$$z = -1.5679713 \quad 20 \frac{dz}{dt} = +0.05416664$$

These values lead to the following perturbed co-ordinates:—

G. M. T.			x	y	z
1898	July	13....	- 3.9803740	+ 0.7339663	- 1.5679713
	June	23....	4.2044239	0.7171655	1.6210429
		3....	4.4232304	0.6994595	1.6720819
	May	14....	4.6372405	0.6809836	1.7212956
	Apr.	24....	4.8468382	0.6618484	1.7688573
		4....	5.0523545	0.6421448	1.8149146
	Mar.	15....	5.2540889	0.6219487	1.8595938
	Feb.	23....	5.4522899	0.6013243	1.9030045
		3....	5.6471875	0.5803266	1.9452424
	Jan.	14....	5.8389844	0.5590028	1.9863921
1897	Dec.	25....	6.0278616	0.5373940	2.0265296
		5....	6.2139831	0.5155363	2.0657228
	Oct.	26....	6.5785402	0.4711939	2.1415169
	Sep.	16....	6.9336942	0.4261793	2.2142024
	Aug.	7....	7.2803209	0.3806469	2.2841321
	June	28....	7.6191655	0.3347153	2.3515993
	May	19....	7.9508669	0.2884762	2.4168488
	Apr.	9....	8.2759793	0.2420035	2.4800880
	Feb.	28....	8.5949862	0.1953550	2.5414930
	Jan.	19....	8.9083123	0.1485781	2.6012166
1896	Dec.	10....	- 9.2163332	+ 0.1017119	- 2.6593910

G. M. T.	x	y	z
1896 Oct. 31	- 9.5193835	+ 0.0547878	- 2.7161315
Sep. 21	9.8177625	+ 0.0078325	2.7715408
Aug. 12	10.1117395	- 0.0391322	2.8257091
July 3	10.4015580	0.0860880	2.8787173
May 24	10.6874388	0.1330196	2.9306380
Apr. 14	10.9695831	0.1799144	2.9815361
Mar. 5	11.2481749	0.2267614	3.0314711
1895 Dec. 16	11.7953637	0.3202778	3.1286603
Sep. 27	12.3302033	0.4135133	3.2225805
July 9	12.8537214	0.5064280	3.3135462
Apr. 20	13.3668048	0.5989955	3.4018256
Jan. 30	13.8702286	0.6911966	3.4876483
1894 Nov. 11	14.3646750	0.7830189	3.5712130
Aug. 23	14.8507477	0.8744537	3.6526935
June 4	15.3289852	0.9654958	3.7322420
Mar. 16	15,7998699	1.0561418	3.8099939
1893 Dec. 26	16.2638365	1.1463891	3.8860687
Oct. 7	16.7212784	1.2362363	3.9605741
July 19	17.1725529	1.3256820	4.0336066
Apr. 30	17.6179854	1.4148261	4.1052531
1892 Nov. 21	18.492489	1.591596	4.244696
June 14	19.346884	1.766834	4.379456
Jan. 6	20.182939	1.940422	4.509993
1891 July 30	21.002159	2.112340	4.636698
Feb. 20	21.805827	2.282577	4.759908
1890 Sep. 13	22.595045	2.451126	4.879910
Apr. 6	23,370768	2.617996	4.996957
1889 Oct. 28	24.133827	2.783206	5.111272
May 21	24.884957	2.946793	5.223048
1888 Dec. 12	25.624805	3.108802	5.332460
July 5	26.353957	3.269292	5.439662
Jan. 27	27.072941	3.428330	5.544792
1887 Aug. 20	27.782241	3.585990	5.647975
Mar. 13	28.482303	3.742348	5.749324
1886 Oct. 4	- 29.173540	- 3.897486	- 5.848941

G. M. T.	x	y	z
1886 Apr. 27....	29.856341	4.051488	5.946919
1885 Nov. 18....	30.531069	4.204437	6.043342
June 11....	31.198073	4.356416	6.138289
Jan. 2....	31.857683	4.507503	6.231829
1884 July 26....	32.510216	4.657781	6.324029
Feb. 17....	33.155984	4.807318	6.414948
1883 Sep. 10....	33.795284	4.956179	6.504643
Apr. 3....	34.428409	5.104419	6.593163
1882 Oct. 25....	35.055646	5.252079	6.680556
May 18....	35.677272	5.399183	6.766867
1881 Dec. 9....	36.293551	5.545740	6.852136
July 2....	36.904735	5.691738	6.936401
Jan. 23....	37.511055	5.837145	7.019699
1880 Aug. 16....	38.112718	5.981919	7.102065
Mar. 9....	38.709904	6.126002	7.183530
1879 Oct. 1....	39.302761	6.269335	7.264126
Apr. 24....	39.891415	6.411861	7.343884
1878 Nov. 15....	40.475960	6.553532	7.422832
June 8....	41.056474	6.694311	7.500997
1877 Dec. 30....	41.633020	6.834176	7.578406
July 23....	42.205647	6.973122	7.655083
Feb. 13....	42.774405	7.111148	7.731053
1876 Sep. 6....	43.339337	7.248303	7.806338
Mar. 30....	43.900489	7.384592	7.880959
1875 Oct. 22....	44.457911	7.520064	7.954936
May 15....	45.011650	7.654769	8.028288
1874 Dec. 6....	45.561767	7.788758	8.101034
June 29....	46.108323	7.922090	8.173190
Jan. 20....	46.651385	8.054823	8.244773
1873 Aug. 13....	47.191028	8.187019	8.315796
Mar. 6....	47.727332	8.318737	8.386275
1872 Sep. 27....	48.260387	8.450037	8.456223
Apr. 20....	48.790289	8.580972	8.525653
1871 Nov. 12....	49.317141	8.711593	8.594575
June 5....	49.841056	8.841938	8.663002

On November 12, 1871, when the comet had a distance of 50.8 units from the Sun, it was so remote from the planets that the perturbations had only insignificant influence on the movement of the comet.

Now we compute the following velocity components $\frac{dx}{dt}$, $\frac{dy}{dt}$, $\frac{dz}{dt}$ and reductions to the centre of gravity of the Sun and the 8 major planets ξ_{\odot} , η_{\odot} , ζ_{\odot} , $\frac{d\xi_{\odot}}{dt}$, $\frac{d\eta_{\odot}}{dt}$, $\frac{d\zeta_{\odot}}{dt}$. The addition gives the resulting centre co-ordinates and velocities \bar{x} , \bar{y} , \bar{z} , $\frac{d\bar{x}}{dt}$, $\frac{d\bar{y}}{dt}$, $\frac{d\bar{z}}{dt}$.

$x = -49.31714$	$y = -8.71159$	$z = -8.59458$
$\xi_{\odot} = + 14$	$\eta_{\odot} = - 320$	$\zeta_{\odot} = - 1$
$\bar{x} = -49.31700$	$\bar{y} = -8.71479$	$\bar{z} = -8.59459$
$\frac{dx}{dt} = +0.00328353$	$\frac{dy}{dt} = +0.00081548$	$\frac{dz}{dt} = +0.00042916$
$\frac{d\xi_{\odot}}{dt} = + 566$	$\frac{d\eta_{\odot}}{dt} = + 173$	$\frac{d\zeta_{\odot}}{dt} = - 10$
$\frac{d\bar{x}}{dt} = +0.00328919$	$\frac{d\bar{y}}{dt} = +0.00081721$	$\frac{d\bar{z}}{dt} = +0.00042906$

From these we find:—

$$\bar{r} = \sqrt{\bar{x}^2 + \bar{y}^2 + \bar{z}^2} = 50.81320$$

$$V^2 = \left(\frac{d\bar{x}}{dt}\right)^2 + \left(\frac{d\bar{y}}{dt}\right)^2 + \left(\frac{d\bar{z}}{dt}\right)^2 = 0.00001167070.$$

If these values are substituted in the equation of conservation of energy:—

$$\frac{V^2}{k^2(1+m)} = \frac{2}{\bar{r}} - \frac{1}{\bar{a}},$$

in which:—

$$k^2(1+m) = 0.0002963093,$$

we find:—

$$\frac{1}{\bar{a}} = -0.0000270 \pm 0.0000184.$$

Thus the computation has given the result that the original orbit was hyperbolic, though only in a slight degree.

The result of the investigations up to now is that in 22 out of 23 cases the orbits have changed in a hyperbolic direction during the time when the comets moved from far off to the region of perihelion. Only in 3 cases the computations have shown an original hyperbolic orbit, but two of the results must be considered inconclusive as all perturbations had not been taken into account. The third case is comet 1899 I, where the original $\frac{1}{a}$ is negative, and numerically larger than the mean error.

In fact we have here the first example of a comet for which all perturbations have been taken into account and which all the same shows an originally hyperbolic orbit. However, since a change in $\frac{1}{a}$ equal to 1.5 times the computed mean error would make the orbit elliptical, it cannot be said that an originally hyperbolic orbit has been established.



